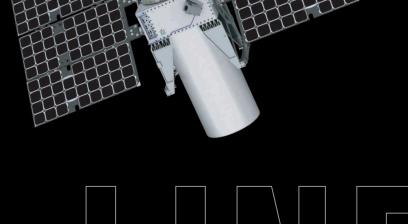


VENUS-100 MICROSATELLITE



CLASS

ESPA- Standard or large 15" launch vehicle interface

POINTING ACCURACY

±0.002° (1-sigma), 3 axes, 2 Trackers

SOLAR ARRAY POWER (BOL)

Two wing: 444 W One wing: 222 W

PAYLOAD MASS CAPABILITY

70 kg

ENERGY STORAGE

13.6 Ah

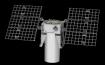
ORBIT ALTITUDE / ORBIT LIFETIME

LEO (> 5 years), GEO (> 2 years), Deep Space (> 2 years)

AVAILABLE PAYLOAD VOLUME

20.5" X 16.4" X 27.0" (1 array) 17.0" X 16.4" X 27.0" (2 array) Larger volume available depending on launch vehicle

MICROSAT & MINISAT SUMMARY





VENUS-100

SATURN-200

CLASS	ESPA-Standard or large 15" launch vehicle interface	ESPA-Grande or Equivalent 24" launch vehicle interface standard, other options available
POINTING ACCURACY	±0.002° (1-sigma), 3 axes, 2 Trackers	
SOLAR ARRAY POWER(BOL)	One wing: 222 W Two wing: 444 W	1082 W
PAYLOAD MASS CAPABILITY	70 kg	Standard Rideshare: 200 kg Specialty applications and options: up to 350 kg *CG and MOI dependent
ENERGY STORAGE	13.6 Ah	54.4 Ah
ORBIT ALTITUDE / ORBIT LIFETIME	LEO (> 5 years), GEO (> 2 years), Deep Space (> 2 years)	
PAYLOAD VOLUME	20.5" X 16.4" X 27.0" (1 array) 17.0" X 16.4" X 27.0" (2 array) Large volume available depending on launch vehicle	30.0" X 30.0" X 40.0" (typical) Larger volume available within rideshare envelope and in dedicate launch vehicle fairings
PROPULSION	Multiple electric and chemica	al propulsion systems available

Note: This data is for information only and subject to change. Please contact Blue Canyon Technologies for current design data.

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